Question.

A 2000 kg satellite orbits the earth at a height of 300 km. What is the speed of the satellite and its period? TakeG=6.67×10-11Nm2/kg2, Mass of the earth is 5.98×1024kg. Given:

$$m = 2000 \ kg$$

$$h = 300 \ km = 3 \cdot 10^5 \ m$$

$$G = 6.67 \cdot 10^{-11} \frac{N \cdot m^2}{kg^2}$$

$$M = 5.98 \cdot 10^{24} \ kg$$

Find:

$$v = ?$$

$$T = ?$$

Solution.

If the satellite moves in circular motion, then the net centripetal force acting upon this orbiting satellite is given by the relationship:

$$F_{net} = \frac{mv^2}{r}$$

This net centripetal force is the result of the gravitational force that attracts the satellite towards the central body and can be represented as:

$$F_{grav} = G \, \frac{mM}{r^2}$$

The above expressions for centripetal force and gravitational force can be set equal to each other. Thus,

$$\frac{mv^2}{r} = G \frac{mM}{r^2}$$

Therefore,

$$v = \sqrt{\frac{GM}{r}}$$

In our case, r = R + h, where $R = 6.37 \cdot 10^6 m$. That's why,

$$v = \sqrt{\frac{GM}{R+h}}$$

By definition

$$T = \frac{2\pi}{\omega}$$

And

$$\omega = \frac{v}{r} = \frac{\sqrt{GM}}{(R+h)^{\frac{3}{2}}}$$

So,

$$T = \frac{2\pi (R+h)^{\frac{3}{2}}}{\sqrt{GM}}$$

Calculate:

$$v = \sqrt{\frac{GM}{R+h}} = \sqrt{\frac{6.67 \cdot 10^{-11} \cdot 5.98 \cdot 10^{24}}{6.67 \cdot 10^6}} = 7730 \frac{m}{s}$$
$$T = \frac{2\pi (R+h)^{\frac{3}{2}}}{\sqrt{GM}} = \frac{6.28 \cdot (6.67 \cdot 10^6)^{\frac{3}{2}}}{\sqrt{6.67 \cdot 10^{-11} \cdot 5.98 \cdot 10^{24}}} = 5420 s$$

Answer.

$$v = \sqrt{\frac{GM}{R+h}} = 7730 \frac{m}{s}$$
$$T = \frac{2\pi (R+h)^{\frac{3}{2}}}{\sqrt{GM}} = 5420 s$$

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