

A satellite has a mass of 6403 kg and is in a circular orbit  $4.42 \times 10^5$  m above the surface of a planet. The period of the orbit is 2.3 hours. The radius of the planet is  $4.73 \times 10^6$  m. What would be the true weight of the satellite if it were at rest on the planet's surface?

### Solution

The height of orbit  $H$ . The radius of the planet -  $R$ . Distance between the planet and the satellite  $H + R$ .

The centripetal force  $F_c$  is provided by the force of gravity  $F_g$  :

$$F_g = F_c \rightarrow \frac{GMm}{(H + R)^2} = m\omega^2(H + R),$$

where  $G$  – constant of gravity,  $M$  - the mass of the planet,  $m$  - mass of satellite,  $\omega$  – angular velocity of the orbit.

We know that  $\omega = \frac{2\pi}{T}$ , where  $T$  - the period of the orbit.

Now we have

$$\frac{GMm}{(H + R)^2} = \frac{4\pi^2m(H + R)}{T^2} \rightarrow \frac{GM}{(H + R)^2} = \frac{4\pi^2(H + R)}{T^2}.$$

The mass of the planet

$$M = \frac{4\pi^2((H + R)^3)}{T^2G} = \frac{4\pi^2(4.42 * 10^5 + 4.73 * 10^6)^3}{(2.3 \text{ h} * 3600 \frac{\text{s}}{\text{h}})^2 * 6.67 * 10^{-11}} = 1.19 * 10^{24} \text{ kg}.$$

The gravitational force,  $F_g$ , is also equal to  $m * g$ . To find the acceleration,

$$mg = \frac{GMm}{R^2} \rightarrow g = \frac{GM}{R^2}.$$

At the surface of the planet,

$$g = \frac{6.67 * 10^{-11} * 1.19 * 10^{24}}{(4.73 * 10^6)^2} = 3.55 \frac{\text{m}}{\text{s}^2}.$$

So,

$$F_g = mg = 6403 * 3.55 = 23 \text{ kN}.$$

**Answer: 23 kN.**