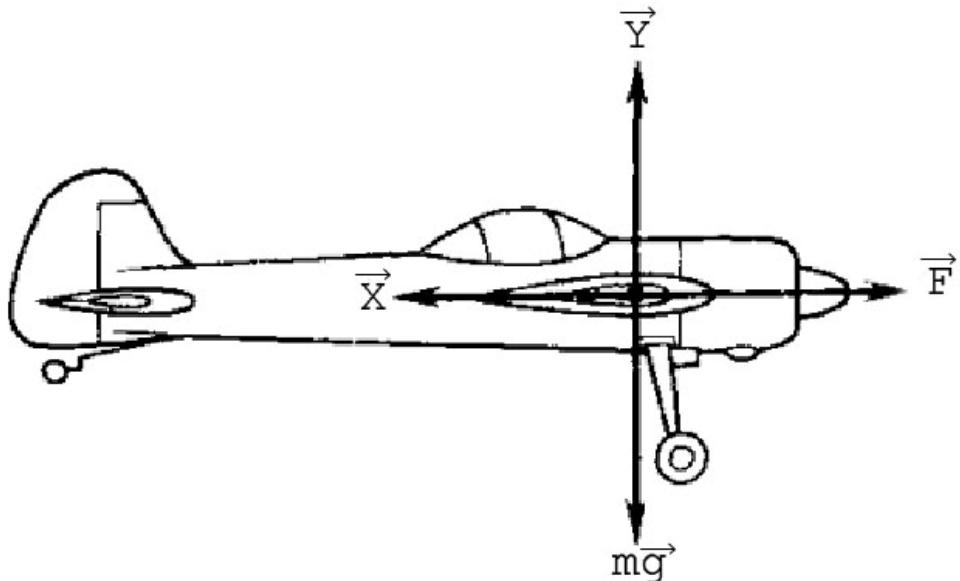


What will happen to the speed of aircraft if the mass of the same reduces during the flight but the engine force remains the same?

**Solution.**



From the theory aerodynamics of the flight: an aircraft flies horizontally, when:

$$Y = mg;$$

$$X = F.$$

in these equations:

$Y$  – the force of the lift;

$mg$  – a weight of the aircraft;

$F$  – the force of the thrust;

$X$  – a force of the drag is opposed to the thrust.

$$Y = C_L \frac{\rho v^2}{2} S;$$

$$X = C_D \frac{\rho v^2}{2} S;$$

in these formulas:

$C_L$  - lift coefficient;

$C_D$  - drag coefficient;

$S$  - a wings area;

$v$  – a speed of the aircraft;

$\rho$  - a density of the air.

We have:

$$m_1 g = C_L \frac{\rho_1 v_1^2}{2} S;$$

$$F = C_D \frac{\rho_1 v_1^2}{2} S;$$

If the mass of the aircraft reduces  $m_2 < m_1$ , then:

$m_2g < C_L \frac{\rho_1 v_1^2}{2} S$  – an aircraft climbs up, where the density of the air is less  $\rho_2 < \rho_1$  (if altitude increases then the air density decreases) and  $m_2g = C_L \frac{\rho_2 v_2^2}{2} S$ .

The engine force remains the same, but  $\rho_2 < \rho_1$ , then the speed of aircraft increases

$v_2 > v_1$  until a force of the drag is equal the force of the thrust  $F = C_D \frac{\rho_2 v_2^2}{2} S$ .

The altitude changes until:

$$m_2g = C_L \frac{\rho_2 v_2^2}{2} S$$

and

$$F = C_D \frac{\rho_2 v_2^2}{2} S.$$

Therefore, if the mass of the aircraft reduces during the flight but the engine force remains the same, the speed of the aircraft increases.

**Answer:** if the mass of the aircraft reduces during the flight but the engine force remains the same, the speed of the aircraft increases.